Interface stress control mechanism of engineering plastic gaskets during the low-amplitude and long-pulse-width dynamic loading process

Feng Xu*, Jianwei Jiang, Jianbin Men, Shuyou Wang and Mei Li

State Key Laboratory of Explosion Science and Safety Protection, Beijing Institute of Technology, Beijing, China

Impact and protection are hot topics of concern in modern military and civilian fields. However, existing research focuses more on high-speed, high-frequency impact, explosion, and other loading conditions, while studies on the protective mechanisms and stress-deformation processes of materials during long-pulse-width and low-amplitude dynamic impact processes are relatively limited. This work investigates the interfacial stress control mechanism of engineering plastic gaskets on protected components under long-pulse-width loading, and employs the finite element method (FEM) to simulate the impact protection processes of gaskets with different material parameters. The influence of gasket material parameters on the interface protection effects for protected vulnerable components was obtained, and the relevant protective mechanisms were revealed. The surface fitting method was used to optimize the performance parameters of protective materials, and the impact protection effect of optimized gaskets under simulated launch loading was verified through large-scale drop hammer tests. Key findings include: (1) Under long-pulse, low-amplitude dynamic loading, the Von Mises stress distribution on impact surfaces of vulnerable components correlates with the anisotropic deformation capacity of protective materials; (2) Controlling the variation and coupling of triaxial stresses at the interface between protective materials and vulnerable components is critical for achieving protection; (3) Engineering plastic protective materials with moderate de-formation capabilities demonstrate superior effects in improving interface stress distribution of typical composite vulnerable components. The research outcomes provide important references for impact protection design under long-pulse dynamic loading process.

KEYWORDS

long-pulse-width dynamic loading, interface impact protection, engineering plastics, finite element method, simulated impact test

1 Introduction

Impact and protection have emerged as critical concerns in modern military and civilian applications. Although research on protective materials has expanded from conventional engineering materials (e.g., polycarbonate, polyethylene, polyurethane elastomers, soft rubber) to composite and structural materials such as fiber-reinforced materials, surface-engineered materials, sandwich materials, gradient materials, and biomimetic materials. Compared with composite and structural materials, traditional engineering protective materials offer the advantages of mature processing technologies, stable mechanical-thermal-chemical properties, and superior cost-effectiveness ratio. These properties make them widely used in military and civilian products.

Nowadays, certain efforts devoted to conventional impact protection research predominantly focus on the safety and protection of critical vulnerable components under high-amplitude, high-frequency impact and explosive loading conditions. For instance, Mohammad (Mohammad and Mohammadzadeh Gonabadi, 2019) employed small-scale drop hammer tests to evaluate the impact resistance of aluminum foam materials, concluding that aluminum foam is suitable as an impact energy-absorbing material. In low-velocity impact tests, Liu Y. R. et al. (2024) found that artificial cartilage materials provide effective cushioning and energy absorption capabilities, which significantly reducing surface stress on aluminum alloys. Furthermore, Sun et al. (2025) conducted high-velocity ballistic impact tests on biomimetic borosilicate glass composites. The results reveal that structural modifications enhance energy dissipation during impact events, thereby improving the protective performance of materials. Similarly, such efforts are reflected in current impact protection research within the military munitions field. For instance, in studies concerning projectile-target plate impact protection: Li et al. (2021) discovered that the stress attenuation capacity of gasket materials is the main factor in determining their protective performance, by compared the impact protection capabilities of different gasket materials using finite element method. However, Li et al. (2017) found the impact protection characteristics of gasket materials are controlled by stress attenuation and dispersion effects, as evidenced by numerical simulations conducted using finite element method. Liang et al. (2020) combined theoretical and experimental methods to analyze projectile penetration processes, demonstrating that foam aluminum-filled thin-walled metal liners effectively reduce impact overload through energy absorption mechanisms. Thus, the dominant factors determining impact protection efficacy remain controversial under identical impact loading conditions due to variations in loading characteristics and protective material properties. Whether in military or civilian applications, it is essential to explore the impact of protective mechanisms and stress-deformation processes on the efficacy of impact protection.

As a distinct form of impact loading, projectiles load launched by artillery exhibit launch overload peaks compared with projectile-target impact loading. These peaks typically range from several thousand to over ten thousand g, with pulse durations of approximately 10 milliseconds. However, projectile-target impacts generate overload magnitudes of tens of thousands of g, characterized by pulse widths spanning several to tens of

micro-seconds (Jia et al., 2023; apps.dtic.mil, 2025). Therefore, compared to the projectile-target impact process, the impact during projectile launch constitutes a typical low-amplitude, long-pulse-duration dynamic loading. Under such loading conditions, the motion characteristics of projectile-borne components resemble mechanical vibrations under high accelerations. However, there is currently no research on the protective effects of common engineering gasket materials on vulnerable components under projectile launch loading.

Based on this, this work employs the finite element method to establish simplified models and calculate the impact protection process of traditional energetic composite gaskets under low-amplitude, long-pulse-duration dynamic loading. The influence of gasket material parameters on the surface/interface protection of vulnerable components is investigated, with the axial loading of projectile launch as the typical external load. Further mechanistic studies identify that the relevant protection mechanisms and optimal material ranges. Finally, experimental verification is conducted with selected typical engineering plastic gaskets.

2 Finite element simulation model and verification

Existing formulas can be used to calculate the axial stress σ_z at the impact interface of protected vulnerable components under launch overload, but the stress distribution cannot be determined. Therefore, this article explores the role of gaskets based on numerical simulation methods and verifies the effectiveness of the simulation calculation model using classical theoretical formulas.

2.1 Finite element model

In the internal ballistic simulation, this study rationally simplifies the projectile structural calculation model, as shown in Figure 1. The simplified model consists of the protected vulnerable component, the gasket bonded to its base, and the simplified projectile casing structure. In this model, D_m denotes the projectile diameter, D represents the diameter of both the vulnerable component and gasket, L_e is the vulnerable component length, H_p is the gasket thickness, and H_M the equivalent thickness at the projectile casing base. A gap of 0.02%D is configured between the vulnerable component and circumferential projectile casing to simulate actual assembly clearances. During launch simulation, pressure load is applied to the projectile casing base, inducing impact loading throughout the entire model.

The TRUE GRID software was employed to establish the corresponding finite element model, as illustrated in Figure 2. The computational structure utilizes an axisymmetric model, with a two-dimensional axisymmetric approach adopted to optimize computational efficiency. To maintain consistency with the dimensions of vulnerable component in the experimental action and to amplify the influence of different gasket materials for enhanced differentiation of the result, the finite element model incorporates specific dimensional parameters: a vulnerable component size of D

FIGURE 1
Simplified model of charge structure with gasket.
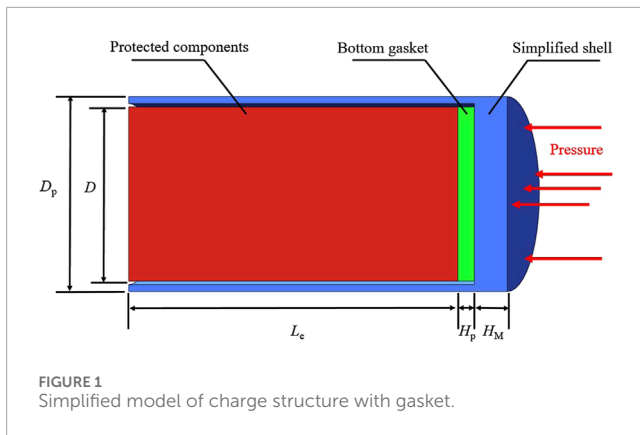
= 40 mm and L_e = 40 mm, and a significantly thick gasket layer of H_p = 10 mm.

All external surfaces of the components are designed as free collision boundaries, and symmetric constraints are imposed on the symmetry plane. In actual physical assembly relationships, the adhesive is used to tightly fit vulnerable components and gaskets. This bonding fit is not easily damaged under axial stress load, resulting in synchronous deformation of vulnerable components and gaskets at the bonding interface. Therefore, shared nodes are used between vulnerable components and gaskets in the finite element calculation of this article to simulate the bonding surface of the actual interface. Meanwhile, the load width is usually 6–13 m according to the load characteristics during the firing process of artillery projectile. The input load waveform is set as a triangular wave, as shown in Figure 2, to simulate the chamber pressure overload with a load time of 10 m. To facilitate the collection and statistical analysis of data, establish 20 observation points along the bottom interface of the vulnerable component, arranged sequentially from the axis of symmetry outwards (i.e., the center of the bottom surface of the vulnerable component is the first observation point and the outer edges are the 20 the observation points).

2.2 Material constitutive model and parameters

2.2.1 The constitutive model of gasket material

Considering strain rate effects and deformation magnitude, the gasket material adopts a kinematic hardening combined elastic-plastic constitutive model in the calculation. As the present computational analysis covers a relatively broad range of material parameters, this section provides the parameter ranges of the material model used in the calculations, as listed in Table 1. The specific elastic modulus E and Poisson's ratio μ adopted in the actual computations will be specified in subsequent sections. Additionally, the influence of gasket material density (ρ) on calculation results is negligible, as demonstrated by theoretical formulas and simulation results in Section 2.3. Therefore, except for the three typical materials specified in Section 2.3 for model validation, all other calculations use a fixed density of 1.5 g/cm^3.

2.2.2 The constitutive model of shell

In the present computational analysis, the shell material is specified as steel. Based on the same considerations, the constitutive model employed for the shell material in the calculations is tabulated in Table 2.

2.2.3 The constitutive model of protected component

Considering the requirements for facilitate experimental validation in Section 4.2 and the operational status of actual projectile-borne vulnerable components, this study configures the vulnerable components with typical projectile-borne energetic composite material. This material is characterized by a generalized Maxwell viscoelastic constitutive model, as shown in Figure 3.

The generalized Maxwell visco-elastic model consists of K visco-elastic elements. The relationship between deviatoric strain and deviatoric stress of visco-elastic elements as Equation 1.

$$S_{ij}(t) = \int_0^t 2G^*(t - \zeta)\dot{e}_{ij}^{ve}d\zeta \tag{1}$$

$$G^*(t) = G^\infty + \sum_{k=1}^N G^{(k)} \exp\left(\frac{-t}{\tau^{(k)}}\right) \tag{2}$$

In Equation 2, $G^{(k)}$ and $\tau^{(k)}$ represent the shear modulus and relaxation time of the Kth Maxwell body, respectively; G^∞ is the shear modulus at infinite relaxation time.

In the finite element computations, analogous formulation mechanical parameters are employed as substitutes, as tabulated in Table 3. This generalized Maxwell viscoelastic constitutive model for the vulnerable composite materials is implemented in the ANSYS/LS-DYNA explicit dynamics solver for computational analysis via secondary development methodology.

2.3 Meshing sensitivity analysis

Analyze the meshing sensitivity for a gasket with an elastic modulus of E = 800 MPa and a Poisson's ratio of μ = 0.4. The mesh element type is quadrilateral shell element mesh. And the calculate results of meshing sensitivity are shown in Figure 4. The X-axis represents the number of grids per millimeter, and the Y-axis represents the maximum Von Mises stress at the bottom layer of protected component. The grid density of 1 grid per millimeter is sufficient to ensure computational accuracy while maximizing computational efficiency. Therefore, the final model was determined to comprise 1880 nodes and 1720 elements.

2.4 Finite element model validation and comparison

Common gasket materials used in engineering include metal, high-molecular engineering plastics, and flexible materials, etc. This article selects steel, low-density polyethylene, and soft rubber as representatives of three different types of materials. These three materials are utilized for model validation and comparative analysis to produce typical calculation results. The parameter values of the three gasket materials are shown in Table 4.
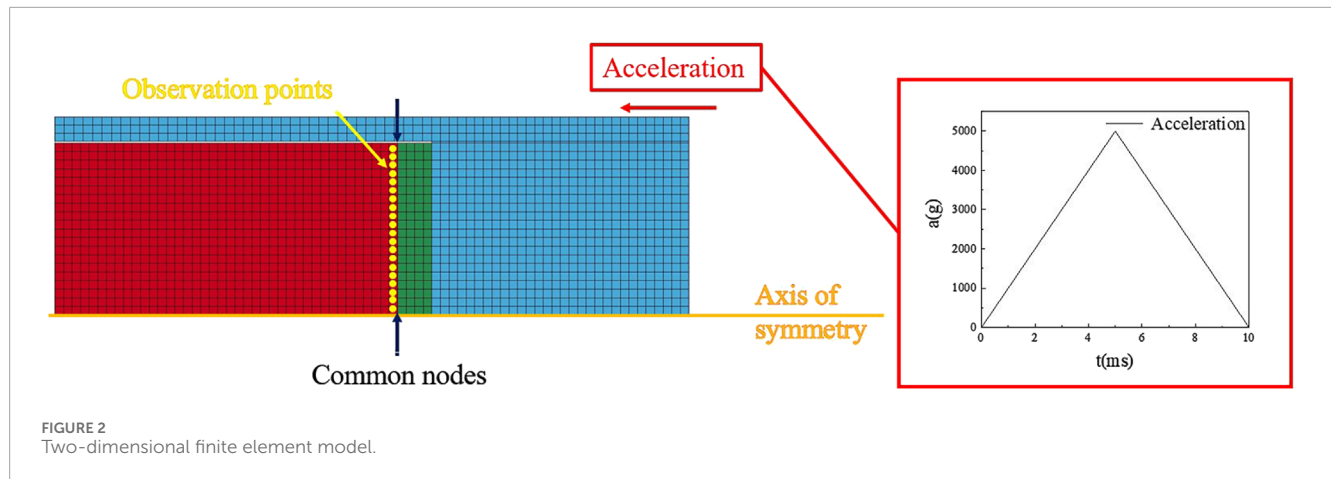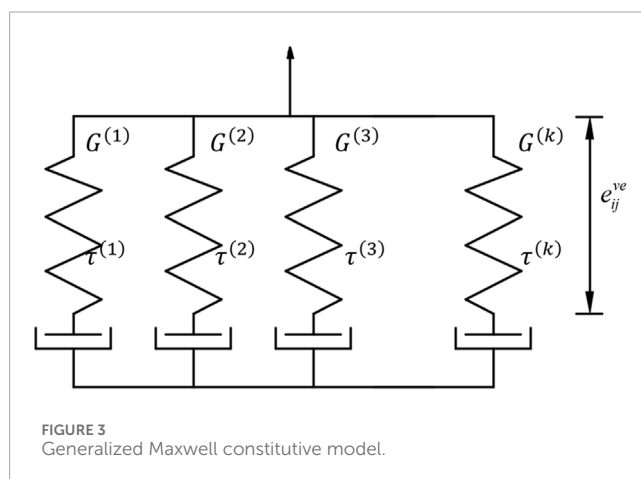
FIGURE 2
Two-dimensional finite element model.

TABLE 1 Parameter range of gasket materials.

ρ(g/cm^3)	E (MPa)	μ	σ_0 (GPa)
1.5	20-3 $\times$ 10^5	0.3–0.49	1.0

TABLE 2 Steel material parameter.

ρ(g/cm^3)	E (MPa)	μ	σ_0 (GPa)
7.85	210 $\times$ 10^3	0.27	1.4

FIGURE 3
Generalized Maxwell constitutive model.

2.4.1 Finite element model validation

According to classical theory, under axial high-g loading the axial stress in the bottom layer of the protected vulnerable component is (Huang, 2014):

$$\sigma_z = \frac{4F_z}{\pi(D_{n0}^2 - D_n^2)} = \frac{4am_e}{\pi(D_{n0}^2 - D_n^2)} = \frac{D_p^2 P_m m_e}{M_p(D_{n0}^2 - D_n^2)} \quad (3)$$

Where, "F_z" is the axial force on the bottom surface of the vulnerable component. "a" represents the acceleration of the entire projectile system. "m_e" denotes the mass of the vulnerable

component. "M_p" indicates the projectile mass. "P_m" refers to the single-shot maximum chamber pressure. "D_p" represents the projectile diameter. "D_n" is the outermost diameter of the vulnerable component (in this paper, taken as the vulnerable component diameter D). "D_{n0}" indicates the innermost diameter of the vulnerable component (in this paper, taken as 0).

Since the calculation result of Equation 3 represents the average axial stress on the bottom surface of the vulnerable component, this article collects data from 20 uniformly distributed stress monitoring points on the bottom surface of the vulnerable component under three typical simulation conditions. It calculates the average axial stress and compares it with the theoretical calculation results of Equation 3, as shown in Table 5. The calculation results indicate that the maximum percentage error between the finite element simulation results and the theoretical prediction results in this article are less than 0.8%. This proves that the simulation model is accurate and effective.

### 2.4.2 Comparison of simulation outcomes for typical working conditions

Obtain axial stress data from 20 observation points on the bottom surface of the vulnerable component under three typical simulation conditions, as shown in Figure 5.

As shown in Figure 5, different bottom gasket materials not only affect the magnitude of the bottom layer stress of the vulnerable component, but also affect the axial stress distribution and temporal variation at various monitoring points on the bottom layer of the vulnerable component. More importantly, comparing the stress oscillations in Figures 5b,c, it is found that the compression and release of the gasket interferes with the vibration of the underlying stress of the vulnerable component under impact. This interference results increases in the peak stress of the bottom layer of the vulnerable component. And this effect occurs for gasket materials with better elasticity when overloaded during launch. Therefore, the protection of the vulnerable composite material component by the gasket under the launch load is not achieved through "buffering," "attenuation" or similar effects.

The Figures 6a,b show the Von Mises stress and axial stress contour charts at the moment of maximum stress in the bottom layer of the vulnerable component under three different working conditions.

TABLE 3 Vulnerable composite material parameters based on the generalized Maxwell viscoelastic constitutive model (Liu W. et al., 2024; Zhang et al., 2016).

G_1 (MPa)	G_2 (MPa)	G_3 (MPa)	G_4 (MPa)	G_5 (MPa)	$1/\tau_1$ (s^{-1})	$1/\tau_2$ (s^{-1})	$1/\tau_3$ (s^{-1})	$1/\tau_4$ (s^{-1})	$1/\tau_5$ (s^{-1})
944.0	173.8	521.2	908.5	687.5	0	7.32×10^3	7.32×10^4	7.32×10^5	2×10^6

FIGURE 4
Calculate results of meshing sensitivity.
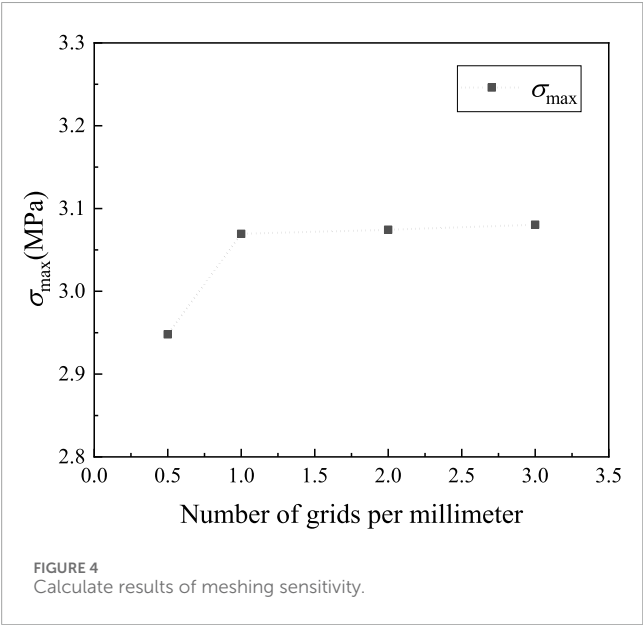
The calculation results indicate that the value and distribution of stress in the bottom layer of the vulnerable component vary among three distinct gasket conditions under the same load. Additionally, it is necessary to distinguish and describe different stresses, such as Von Mises stress and axial stress. Therefore, for the convenience of discussion, all stresses mentioned in this article refer to the corresponding values at the moment of maximum stress occurrence. The parameters are introduced such as $(\sigma)_i$ and σ_{max}, and their definitions are provided in Table 6. During the discussion, subscripts are added to parameter (z for axial stress, r for radial stress, and τ for circumferential stress), whereas Von Mises stress does not have subscript.

The relationship between the peak stress values and distribution positions of $(\sigma_z)_i$ and $(\sigma)_i$ at each monitoring point under three typical working conditions is shown in Figures 7a,b, respectively.

As shown in Figure 7a, the $(\sigma_z)_i$ distribution at the center of the vulnerable component's bottom surface remains relatively consistent across the three working conditions. However, notable differences emerge in $(\sigma)_i$ at the outer edge, primarily due to local stress concentration effects resulting from the deformation of the vulnerable component under stress. This deformation also significantly impacts $(\sigma)_i$ on the bottom surface of the vulnerable component. As shown in Figure 7b, $(\sigma)_i$ in the central area of the bottom surface of the vulnerable component is consistently low across all three working conditions. In contrast, $(\sigma)_i$ on the outer edge is generally higher. Among them, $(\sigma)_i$ in working condition 3 is consistently higher than that in working conditions 1 and 2. Although $(\sigma)_i$ under condition 1 is generally lower than that under condition 2, the peak value on the outer side is higher.

3 Simulation calculation and analysis

Further analyze the influence of E and μ of the gasket material on the stress distribution on the bottom surface of the vulnerable component to reveal the underlying mechanism and provide guidance for optimal material selection. This article conducts numerical simulations across 6 groups, totaling 90 working conditions. In the specific calculation conditions, μ is taken as 0.25, 0.3, 0.35, 0.4, 0.45, and 0.49, respectively. At the same time, 15 E values (20, 50, 100, 300, 500, 800, 1,000, 2000, 3,000, 5,000, 1×10^4, 3×10^4, 5×10^4, 1×10^5 and 3×10^5 MPa) are taken for each μ. The overall range of parameter values covers the vast majority of commonly used engineering materials.

3.1 Finite element calculation results

Figures 8, 9 respectively show the variation of the maximum stress and average stress of the bottom layer of the vulnerable component with material parameters. For σ_z, since its sign only indicates the direction, its absolute value is taken here.

As shown in Figure 8b, the maximum stresses value σ exhibits a trend of decreasing initially, followed by the increases as E of the gasket increases. This is roughly consistent with the variation law of σ_z with E of the gasket in Figure 8a. This indicates a strong correlation between the σ and σ_z. Meanwhile, as shown in Figure 8, when E of the gasket is small, its μ exhibits a significant positive correlation with the value of σ and σ_z in the bottom layer of the vulnerable component. Conversely, when E of the gasket is large, its μ has minimal influence on σ and σ_z.

For the average stress, as shown in Figure 9a, the change in gasket material has almost no effect on the average σ_z. There is only a slight difference when $E \leq 100$ MPa. This indicates that the deformation ability of the gasket material is enhanced, resulting in a large amplitude oscillation of the overall σ_z. As shown in Figure 9b, the average σ in the bottom layer of the vulnerable component initially decreases and then stabilizes with the increase of E of the gasket, which is significantly different from the trend of the average σ_z.

3.2 Analysis of the protective mechanism of gasket materials on vulnerable components

3.2.1 Stress homogenization and deformation equilibrium point
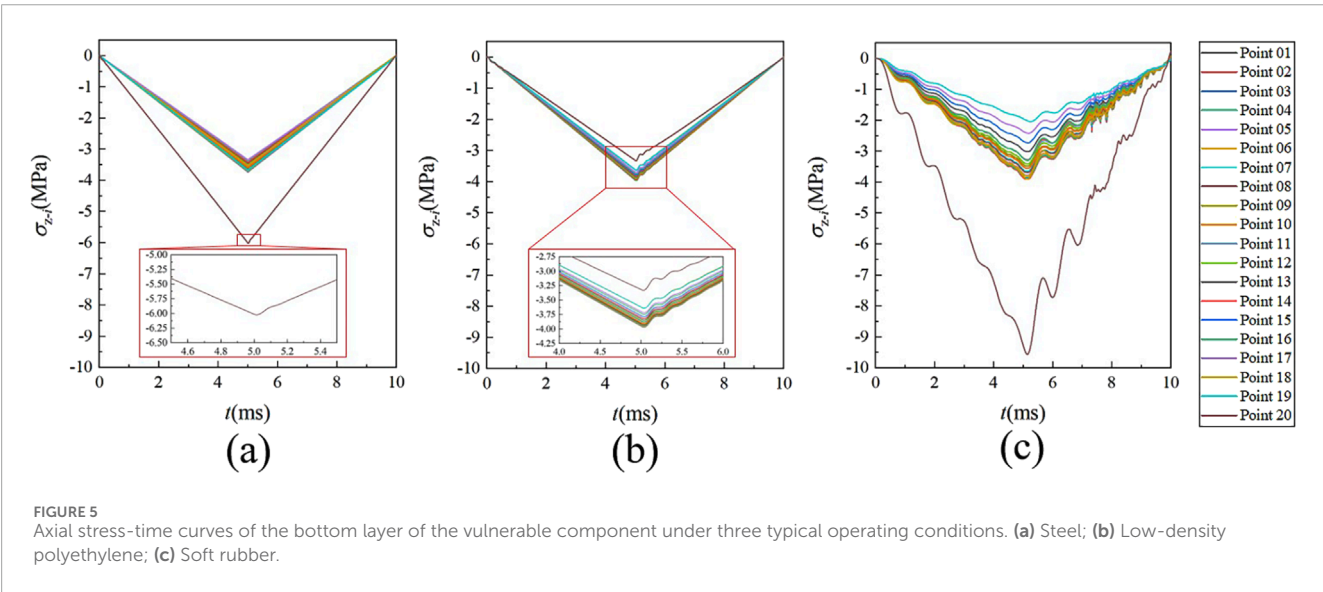
As discussed in Section 3.1, significant changes occur in the regularity of force distribution on the bottom layer of the vulnerable component when E of the gasket ranges from 100 to 2000 MPa. Therefore, this E range is crucial for revealing the underlying mechanism. Meanwhile, the difference in regularity between axial

TABLE 4 Typical gasket material parameters for model validation.

Working condition	Material	ρ(g/cm^3)	E (MPa)	μ
1	Steel	7.85	2.1×10^5	0.27
2	Low-density polyethylene	0.95	7.22×10^2	0.4
3	Soft rubber	1.15	20	0.475

TABLE 5 Comparison between simulated and theoretical axial stress results under typical working conditions.

Working condition	Simulation result (MPa)	Theoretical result (MPa)	Percentage error (%)
1	−3.762		0.56
2	−3.757	−3.783	0.69
3	−3.813		0.79

FIGURE 5
Axial stress-time curves of the bottom layer of the vulnerable component under three typical operating conditions. (a) Steel; (b) Low-density polyethylene; (c) Soft rubber.

stress σ_z and corresponding Von Mises stress σ in Figures 8, 9 indicates that other factors affect σ, leading to its current state. To better understand the protective mechanism of the gasket on the bottom layer of the vulnerable component, we investigated the variations in the value and distribution of the radial stress σ_r and circumferential stress σ_τ in this layer when E of the gasket ranges from 100 to 2000 MPa.

The working condition with $\mu = 0.4$ is selected as the research object. Figure 10a illustrates the value and distribution of σ_z under various working conditions within this range. The contour charts of σ_z of the vulnerable component under various working conditions within this interval are presented in Figures 10b–g. The symmetrical axis of each contour chart is on the left side of the graph. The interface between the vulnerable component and gasket (i.e., the bottom of the vulnerable component) is positioned in the upper region of the graph. The overall positional relationship is consistent with Figure 10a for comparison purposes.

Firstly, as shown in the axial stress contour charts in Figures 10b–g, within the range of 100–2000 MPa for E of the gasket, the peak area (blue part) of the absolute σ_z of the vulnerable component has undergone a shift. At the same time, according to the information in Figure 7a of Section 2.4.2 and Figure 10a, the absolute value of σ_z at the center is higher than that at the outer edge only when E is in the range of 300–1,000 MPa. And in other ranges, the peak σ_z of the bottom layer is always located at its outer edge.

As described in Section 2.4.2, when the range of E is less than 300 MPa or greater than 1,000 MPa, the stress concentration of vulnerable components is mainly near the outer edge. This is due to the difference in radial deformation between vulnerable components and gaskets under axial force. This results in a sharp change in the local shape of the outer edge of vulnerable components. When the E ranges from 300 to 1,000 MPa, the peak stress in the bottom layer of the vulnerable component is observed at the of its the center of bottom. This phenomenon is more akin to the stress distribution
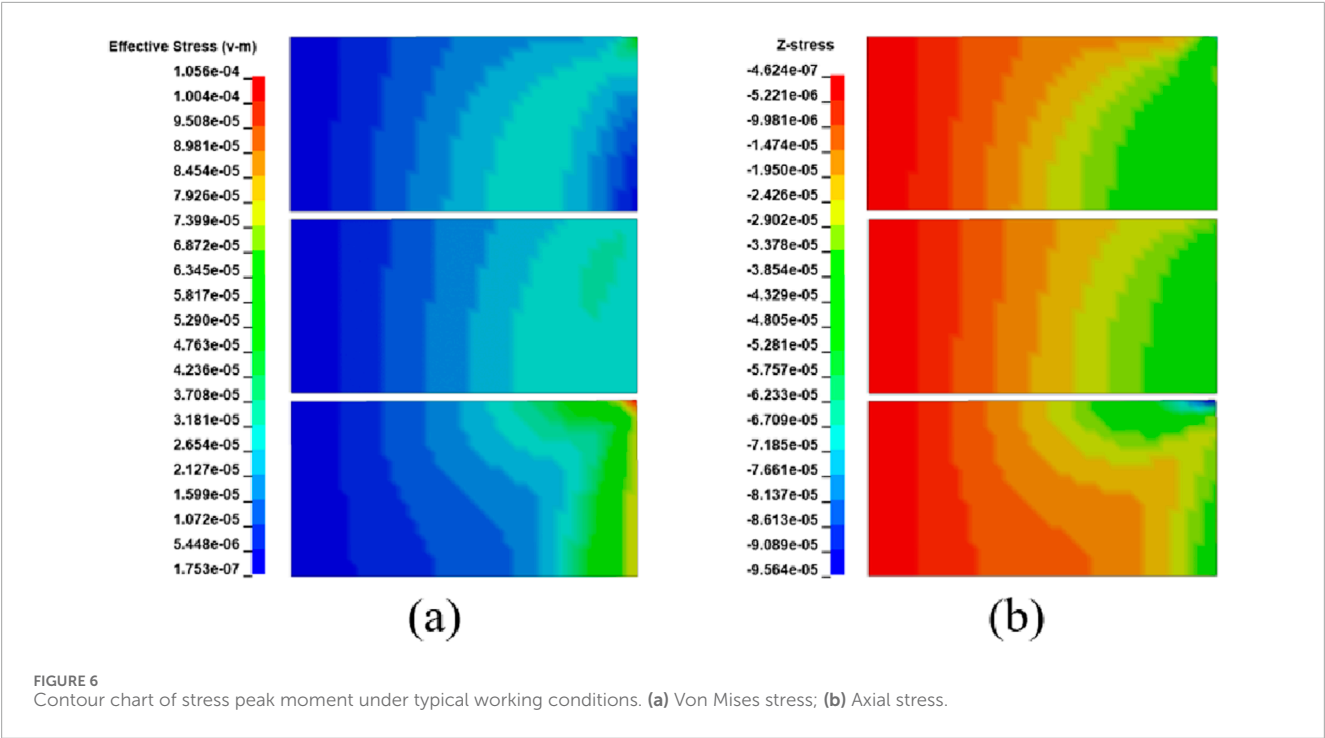
FIGURE 6
Contour chart of stress peak moment under typical working conditions. **(a)** Von Mises stress; **(b)** Axial stress.

TABLE 6 Definitions of parameters for stress distribution.

Parameter	Definition
$(\sigma)_i$	The peak stress recorded at the i-th observation point
σ_{max}	Maximum stress: $\sigma_{max} = \max\{(\sigma)_i\}$
$\bar{\sigma}$	Average stress: $\bar{\sigma} = \sum_{i=1}^{20} (\sigma)_i \times A_i$, A_i is the area of the circular ring contained in the i-th monitoring element
ϕ_{max}	Maximum stress concentration value: $\phi_{max} = \sigma_{max}/\bar{\sigma}$

on the cross-section of an ideal cylindrical structure under axial loading. Therefore, there is reason to suspect that, within E range of the gasket (between 300 and 1,000 MPa), the existence of a point of E that results in no axial stress concentration in the bottom layer of the vulnerable component at this value. This point is the deformation equilibrium point E_{eq} described in this article.

The σ_r state of the vulnerable component supports this hypothesis. Figure 11 shows σ_r state of the vulnerable component when E is in the range of 100–2000 MPa ($\mu = 0.4$), which has similar meaning to that of Figure 10. As shown in Figures 11b–g, when $E \leq$ 300 MPa, σ_r in the bottom layer is positive. This indicates that when E of the gasket is low, the overall radial deformation of the gasket exceeds that of the vulnerable component. This causes the bottom layer of the vulnerable component to become compressed. Based on the data in Figure 11a, a deformation equilibrium point E_{eq} for the gasket material is proven in the range of 300 MPa < E < 500 MPa ($\mu = 0.4$), which makes σ_r zero. At this time, no axial stress is concentrated in the bottom layer of the vulnerable component. Thus, a reasonable gasket material can play a role in homogenizing the bottom stress of the vulnerable component.

3.2.2 Stress coordination function and stress coordination point

First, it is defined that the "stress coordination function" in this paper specifically refers to the mechanism of reducing von Mises stress through controlled variations in triaxial stresses.

Based on the above conclusions, this section further discusses the regularity of Von Mises stress σ. Figure 12 shows σ state ($\mu = 0.4$) of the vulnerable component when E of the gasket is in the range of 100–2000 MPa. Figures 12a–g has a similar meaning to Figures 10a–g. As shown in Figure 12a, the σ distribution in the bottom layer of the vulnerable component is essentially uniform in the elastic modulus range of 300–500 MPa for the gasket material. According to Figure 8b in Section 3.1, within the elastic modulus range of 20–300 MPa, the overall σ increases, accompanied by local stress concentration at the outer edge. In the elastic modulus range of 300–3 × 10^5 MPa, the σ gradually increases from the center towards the outer edge.

We discussed the homogenization issue of Von Mises stress in the bottom layer of vulnerable component within the 300~500 MPa range, and verified that the σ_z distribution of the gasket material at E_{eq} is uniform ($\sigma_r = 0$). Meanwhile, existing theories have shown that an ideal cylinder component experiences circumferential stress $\sigma_r = \sigma_\tau$ (Huang, 2014) under axial overload conditions. This is consistent with the simulation results presented in this paper, which will not be further elaborated. Therefore, combining the σ formula shows that when the gasket material is near E_{eq}, its σ value and distribution are primarily determined by the σ_z value and distribution. This causes the similar stress distribution uniformity characteristics in the same interval. The Von Mises stress formula is shown in Equation 4.

$$\sigma = \left(\sqrt{2}/2\right)\sqrt{(\sigma_z - \sigma_r)^2 + (\sigma_r - \sigma_\tau)^2 + (\sigma_\tau - \sigma_z)^2} \quad (4)$$
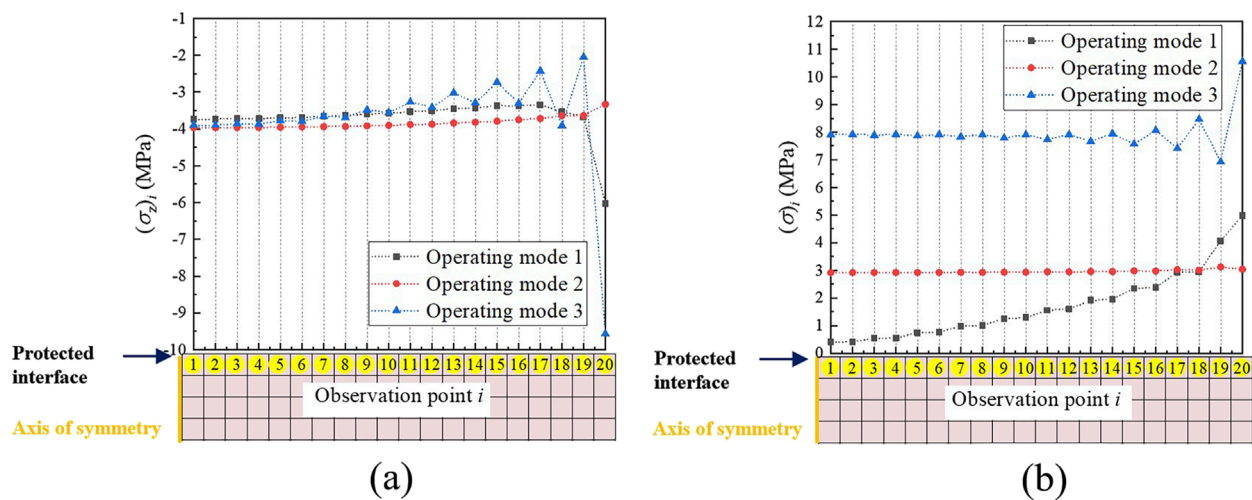
FIGURE 7
Stress distribution in the bottom layer of the vulnerable component under typical working conditions. (a) $(\sigma_z)_i$ distribution; (b) $(\sigma)_i$ distribution.
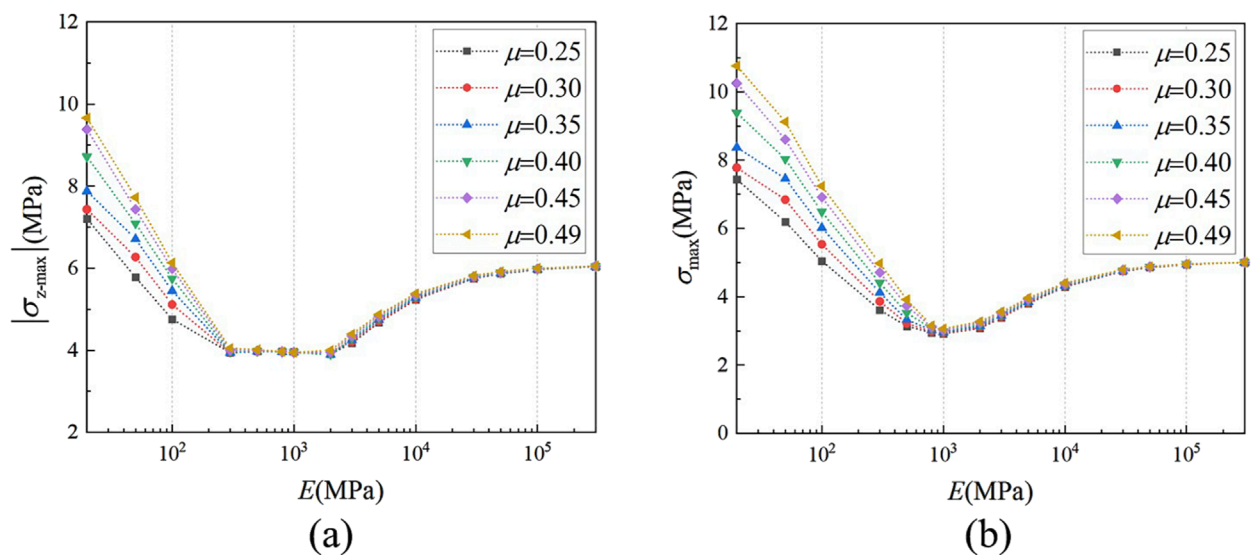

FIGURE 8
The influence of material parameters on maximum stress. (a) $\sigma_{z\text{-max}}$; (b) σ_{max}.

Similarly, when σ not within the 300–500 MPa range, the influence of the distribution of σ_r and σ_τ must be considered. Typically, under ideal conditions $\sigma_r = \sigma_\tau$. Therefore, the variation of vulnerable component deformation, gasket deformation, σ_z and σ_r with E of the gasket were plotted as shown in Figure 13. Among them, the stress coordination point (E_{sc}) denotes E of the gasket corresponding to the point of minimum σ. Although the minimum value of the simulation result in Figure 8b of Section 3.1 corresponds to $E = 1,000$ MPa, the selection of simulation calculation parameters is spaced similarly to E_{eq}. This means that the E_{sc} can only be determined to have a value ranging from 500 to 2000 MPa ($\mu = 0.4$).

As shown in Figure 13, under the condition of gasket $\mu = 0.4$, the pattern of how σ varies with E of gasket is divided into the following five states.

State (1): When E of the gasket is less than or equal to 100 MPa, the deformation of the gasket increases significantly as its E decreases. This results in a concurrent increase in σ_z at the outer edge of the bottom layer of the vulnerable component. Simultaneously, this deformation causes a concurrent increase in σ_r. It should be noted that in this state, σ_z is negative while $\sigma_r = \sigma_\tau$ is positive. Therefore, according to the Von Mises stress formula, σ at the bottom layer of the vulnerable component increases as E decreases.

State (2): When 100 MPa $< E < E_{eq}$, as E approaches E_{eq}, the connection between the gasket and the vulnerable component gradually becomes smoother, resulting to a more uniform distribution of σ_z. Simultaneously, σ continues to decrease as E increases, causing the peak position to shift towards the
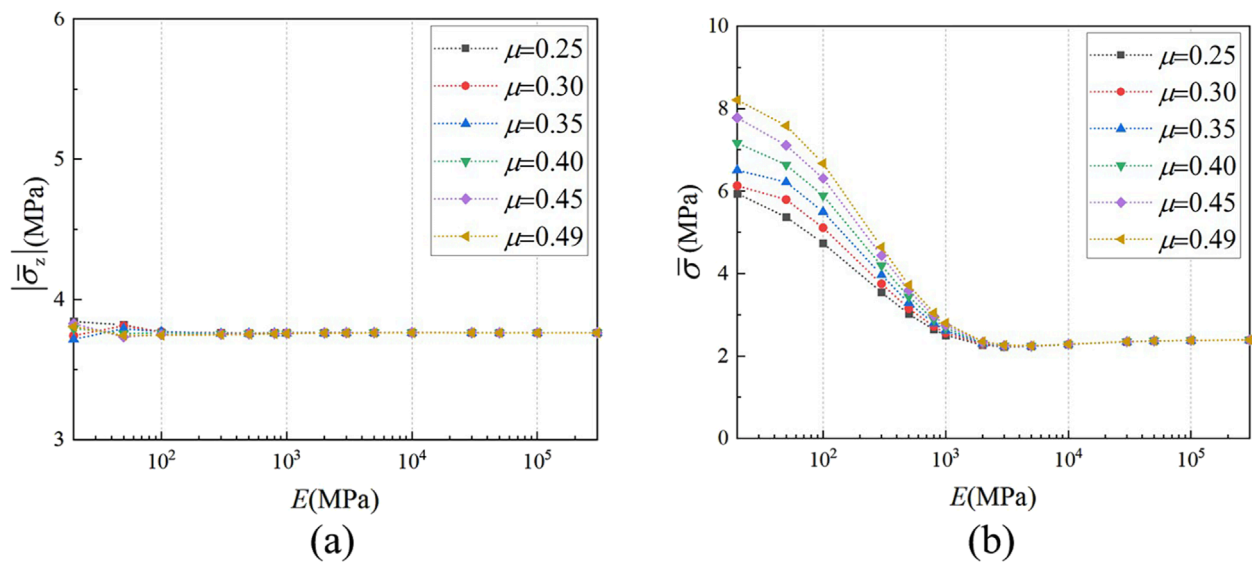
FIGURE 9
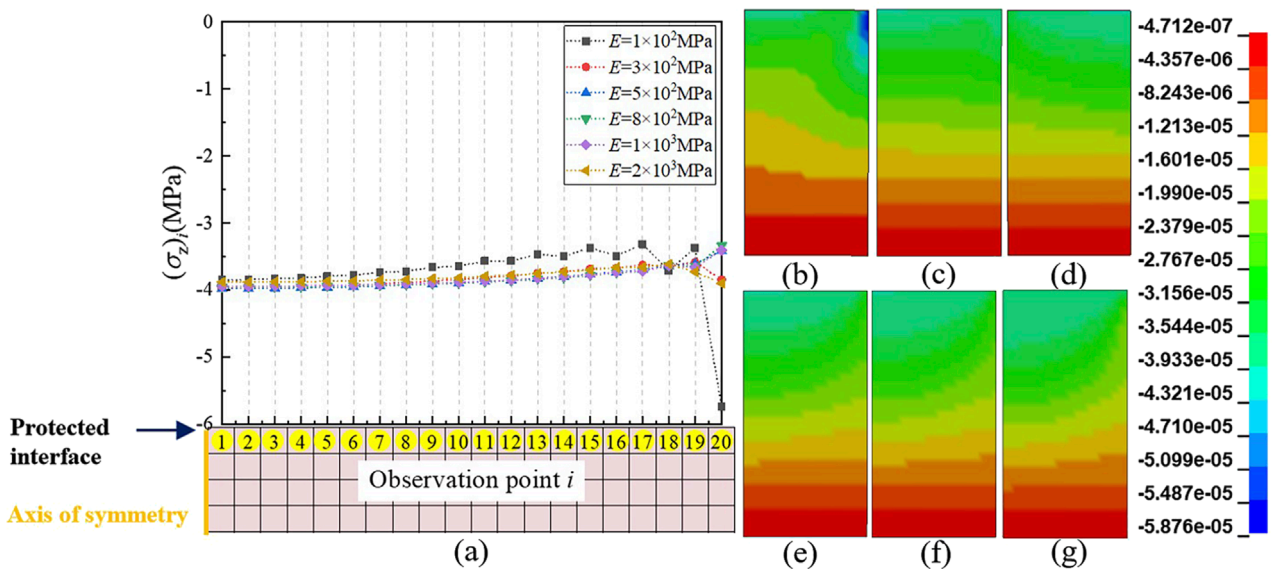The influence of material parameters on average stress. **(a)** $\bar{\sigma}_z$; **(b)** $\bar{\sigma}$.

FIGURE 10
Axial stress state with elastic modulus of 100−2000 MPa (μ = 0.4) for the gasket. **(a)** Axial stress value and distribution; **(b−g)** Contour charts: **(b)** E = 100MPa, **(c)** E = 200MPa, **(d)** E = 500 MPa, **(e)** E = 800 MPa, **(f)** E = 1,000 MPa, **(g)** E = 2000 MPa.

center of the bottom layer of the vulnerable component. At this state, the variation patterns of σ_r and σ_τ are consistent with state (Mohammad and Mohammadzadeh Gonabadi, 2019), and they gradually approach zero as E of the gasket approaches E_{eq}. Therefore, according to the Von Mises stress formula, σ further decreases as E increases.

State (3): When $E = E_{eq}$, the smooth connection between the gasket and the vulnerable component eliminates stress connection, resulting the distribution of σ_z to approximate the ideal stress characteristics of a cylindrical pressure vessel with $\sigma_r = 0$ and $\sigma_\tau = 0$.

At this state, the value and distribution of σ are equal to the absolute value of σ_z.

State (4): When $E_{eq} < E \le E_{sc}$, E of the gasket gradually moves away from E_{eq}. The results in the peak position of σ_z shifting outward from the center to the outer edge of the bottom layer of the vulnerable component. Concurrently, peak stress values transition from initial stability to a gradual increase. However, i the radial deformation of the vulnerable component surpasses that of the gasket at this stage. Thus, both σ_r and σ_τ turn negative. As E of the gasket increases, the value of σ_r and σ_τ gradually rise. According to
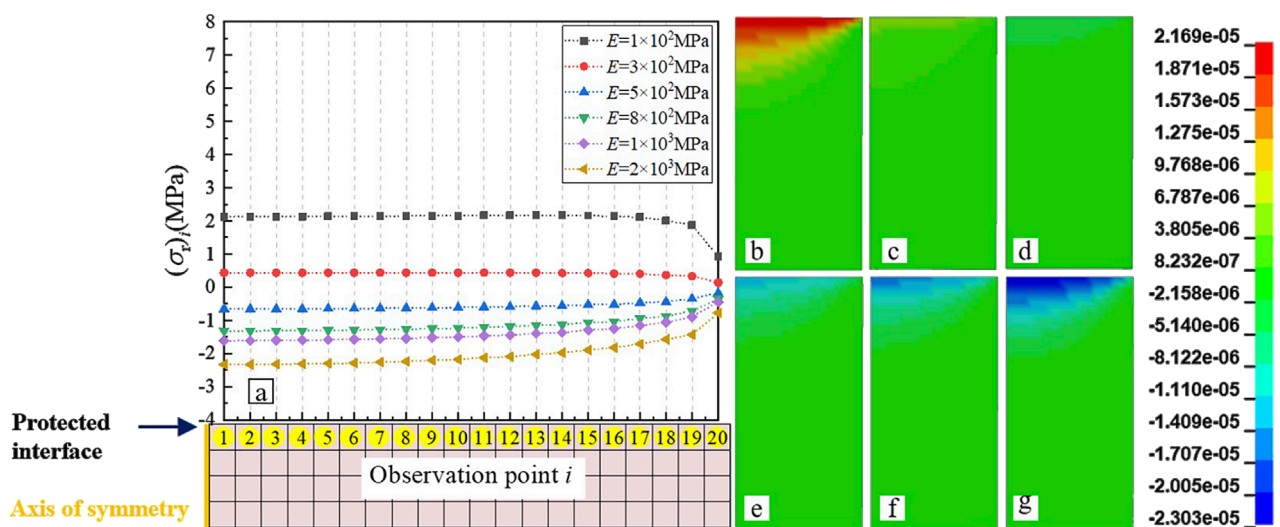
FIGURE 11
Radial stress state with elastic modulus of 100−2000 MPa ($\mu = 0.4$) for the gasket. **(a)** Radial stress value and distribution; **(b−g)** Contour charts: **(b)** E = 100MPa, **(c)** E = 200MPa, **(d)** E = 500MPa, **(e)** E = 800MPa, **(f)** E = 1000MPa, **(g)** E = 2000 MPa.
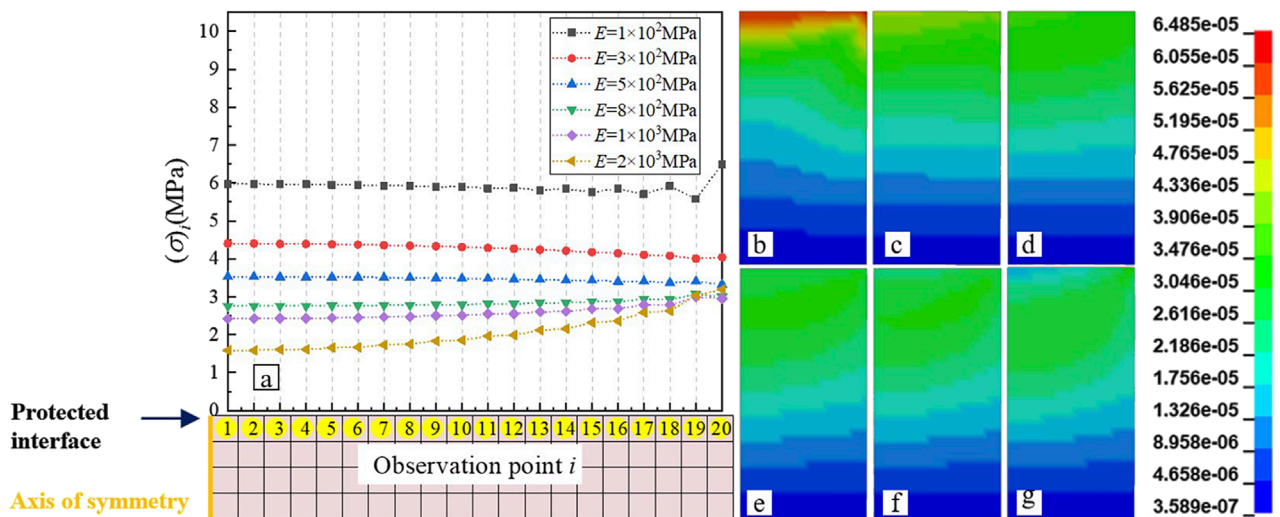

FIGURE 12
Von Mises stress with elastic modulus of 100−2000 MPa ($\mu = 0.4$) for the gasket. **(a)** Von Mises stress value and distribution; **(b−g)** Contour charts: **(b)** E = 100MPa, **(c)** E = 200MPa, **(d)** E = 500MPa, **(e)** E = 800MPa, **(f)** E = 1000MPa, **(g)** E = 2000 MPa.

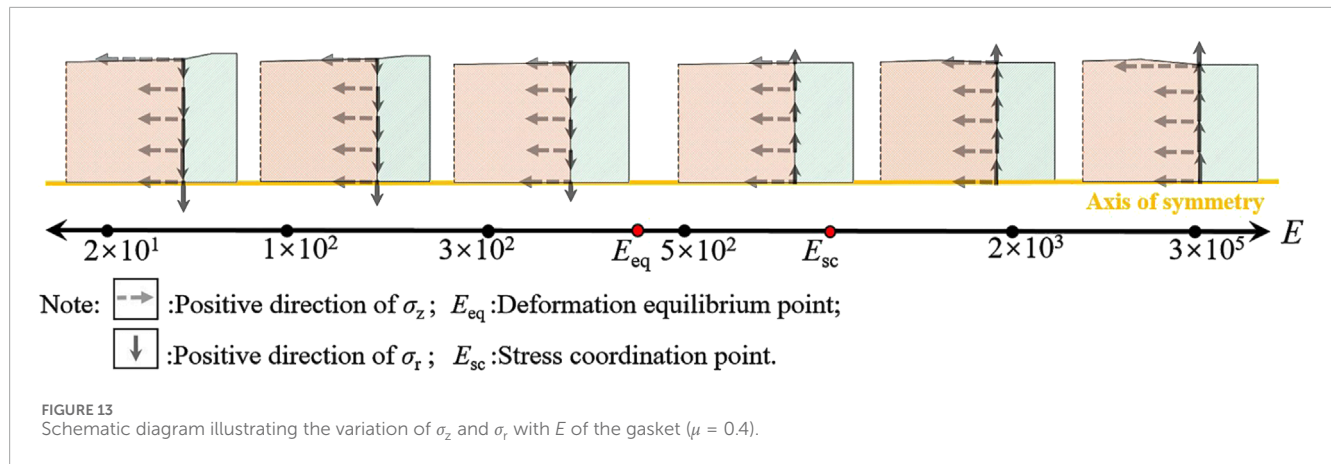Equation 2, σ actually decreases as E increases, gradually reaching its minimum value. This occurs because the presence of negative values σ_r and σ_τ in Equation 2 diminish the dominant influence of the negative value of σ_z on the outcome of formula.

State (5): When $E_{sc} < E$, the deformation capacity of the gasket diminishes as E continues to increase. Concurrently, the deformation of the vulnerable component in relative to the gasket increases and gradually stabilizes at a fixed value. This value corresponds to the deformation of the vulnerable component when the gasket acts as a rigid body. Therefore, as E of the gasket increases, the value of σ_z at the outer edge of the bottom layer also increase, and gradually approaching a fixed value. At the same time, the values

of σ_r and σ_τ also increase synchronously and gradually tend to a fixed value. However, in the distribution of σ_r and σ_τ, the values at the outer edge are the smallest, and their subsequent growth rate is significantly lower than that of σ_z. Therefore, according to the Von Mises stress formula, σ continues to rise as E, gradually approaching a fixed value.

3.2.3 The influence of Poisson's ratio of gasket on the values of deformation equilibrium point and stress coordination point

Theoretically derivation of the calculation formulas for E_{eq} and E_{sc} is challenging due to the adhesive bonding between the bottom

FIGURE 13
Schematic diagram illustrating the variation of σ_z and σ_r with E of the gasket ($\mu = 0.4$).

layer of the vulnerable component and the gasket. However, the correlation between them and μ of the gasket can be discussed through classical material mechanics. Assuming that the vulnerable component and the gasket are both isotropic elastic cylinders, σ_z on both sides of the interface between the gasket and the vulnerable component is the same, σ_r is equal in magnitude and opposite in direction. According to the Poisson effect, the relationship between the radial strain ε_r and axial strain ε_z of the gasket on the bonding interface between the gasket and the bottom layer of the vulnerable component is as Equation 5.

$$\varepsilon_r = -\mu\varepsilon_z \tag{5}$$

Since the σ_z is the same on both sides of the bonding surface between the gasket and the vulnerable component. Therefore, axial strain can be expressed using Equation 6.

$$\varepsilon_z = \sigma_z/E \tag{6}$$

Assuming further that the σ_r and radial displacement u_r on the constraint surface are both related to the geometric dimensions of the vulnerable component. We introduce a shear stress coefficient κ, which is assumed to depend on the material properties and geometric shape of the vulnerable component. The σ_r on the constraint surface can be approximately represented follows.

$$\sigma_r \approx \kappa \cdot u_r = \kappa \cdot \frac{\mu\sigma_z D}{2E} = \kappa \cdot \frac{2\mu ma}{\pi DE} \tag{7}$$

As mentioned earlier in Section 3.2.2, when E of the gasket material is set to E_{eq}, the σ_r is zero. Therefore, we assume that $|\sigma_r|$ is a non-zero minimal value. Based on the left and right limits of the Formula 7, it is not difficult to find that the value of E_{eq} is related to the properties of the vulnerable component material, and geometric shape, and is directly proportional to Poisson's ratio μ of the gasket. E_{sc} is also directly proportional to μ. Figure 14 shows the relationship between material parameters and stress concentration factor. The relationship between E_{eq} and μ cannot be clearly determine due to the oscillation of axial stress in the bottom layer of the vulnerable component's material near E_{eq}, as shown in Figure 14a. However, Figure 14b clearly indicates that as μ of the gasket material increases, so does E of the gasket, This corresponds to the minimum Von Mises stress concentration factor in the bottom

layer of the vulnerable component. This suggests that E_{sc} has shifted in the positive direction of the horizontal axis at this time.

Based on the above discussion, it is established that for given vulnerable component conditions, a specific E_{eq} necessarily exists for the gasket material's elastic modulus. When E is set to this value, σ_z and σ at the bottom layer of the vulnerable component are uniformly distributed. At the same time, there exists a E_{sc} that is slightly greater than E_{eq}. When E is set to this value, the peak of σ achieves its minimum under the combined effects of various stresses. Crucially, E_{eq} and E_{sc} are positively correlated with μ of the gasket material.

3.2.4 The influence of thickness of gasket on the values of deformation equilibrium point and stress coordination point

The thickness variation of the gasket leads to changes in its longitudinal deformation, impacting the homogenization coordination effect. Therefore, further discussion is needed on the relationship between the thickness H_p of the gasket and the homogenization coordination effect. Take typical gasket thicknesses of H_p = 2, 10, 20 mm, and a material Poisson's ratio of μ = 0.4. The calculation results under different elastic modulus E are shown in Figure 15.

As shown in Figure 15, when the elastic modulus of the gasket material is small, the thickness of the gasket will significantly increase the longitudinal deformation difference at the interface between the gasket and the grain. The higher the thickness, the greater the stress concentration factor and stress peak. But when the elastic modulus of the gasket is higher than 300 MPa, the regularity of the three curves is consistent. Meanwhile, the optimal elastic modulus value corresponding to the lowest Von Mises stress suggests that the uniform coordination effect of the gasket on the bottom layer stress of the vulnerable component is independent of the thickness of the gasket.

3.2.4.1 Discussion on the mechanism of action of gasket materials

From the above analysis, the regularity observed in σ is actually a comprehensive reflection of the three-dimensional stress law in the Von Mises stress expression. The minimum value of σ arises from the combined effects of two factors. Firstly, when E of the
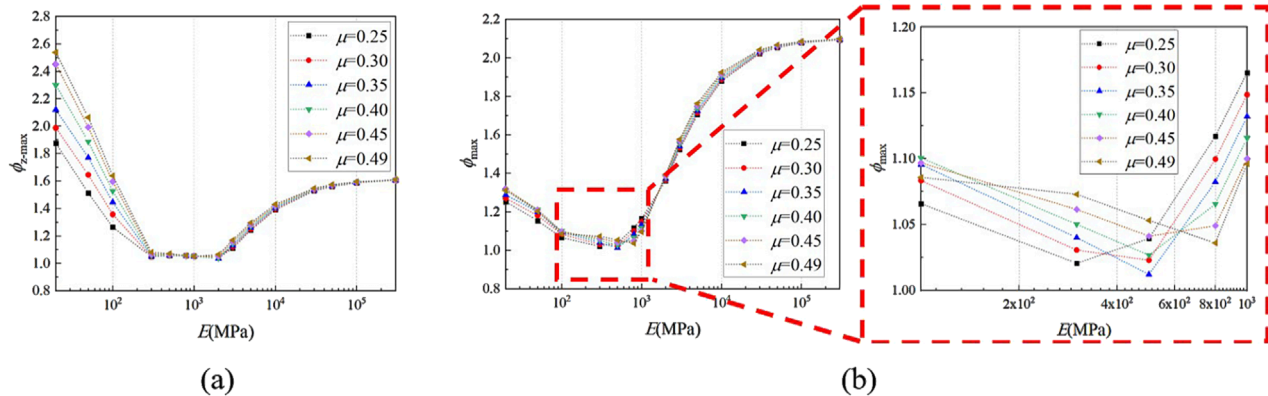
FIGURE 14
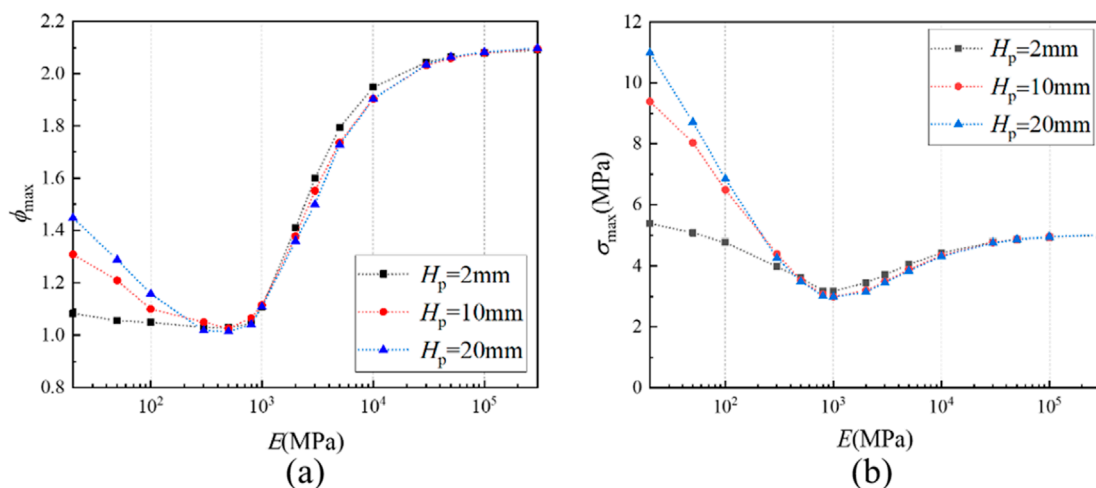The influence of material parameters on stress concentration factor. **(a)** $\phi_{z\text{-max}}$; **(b)** $\phi_{\max}$.

FIGURE 15
The influence of gasket thickness H_p on Von Mises stress in the bottom layer of the vulnerable component. **(a)** $\phi_{\max}$; **(b)** $\sigma_{\max}$.

gasket exceeds E_{eq} and gradually moves away from it, the axial stress σ_z undergoes a slow adjustment in stress distribution. At this time, its value remains relatively low, with the high distribution at the center and low distribution at the edges. Secondly, when E is greater than E_{eq}, σ_r and σ_τ become negative, and their magnitudes increase as E increases. The combined effect of the above two factors results in σ not being the minimum value when the bottom layer of the vulnerable component is under uniform stress distribution at E_{eq}. On the contrary, when E is slightly greater than E_{eq}, the introduction of negative values of σ_r and σ_τ weakens the dominant role of σ_z in Von Mises stress calculation to a certain extent. This can further reduce the calculated value of σ. As E continues to increase, the growth rate at the outer edge of the vulnerable component is much slower than that of σ_z due to their distribution characteristics. Conversely, σ starts to increase again.

Therefore, when E of the gasket is set at E_{eq}, the uniform distribution of stress in the bottom layer of the vulnerable component arises from the consistency of local longitudinal

deformations in both the vulnerable component and the gasket. This phenomenon is defined as the stress homogenization effect of the gasket on the bottom layer of the vulnerable component in this article. When E is set at E_{sc}, the numerical coordinates and coupling of the three-dimensional stresses in the bottom layer of the vulnerable component result in the lowest σ. This effect is referred to the multi-directional stress coordination effect of the gasket on the bottom layer of the vulnerable component. This paper terms these two effects collectively as the homogenization and coordinating effects of the gasket on the bottom interface of the vulnerable component.

3.3 Fast prediction model and gasket material selection

In Section 3.2 uses the example of gasket with $\mu = 0.4$ to analyze and discuss how the gasket homogenizes and coordinates the

TABLE 7 Parameter regression fitting results.

Parameter	Fitting result	Parameter	Fitting result
P_1	-1.3070×10^2	P_6	1.1367×10^1
P_2	9.3016	P_7	6.1084×10^{-3}
P_3	1.8700	P_8	4.7046×10^{-6}
P_4	6.2117×10^{-1}	P_9	3.9430×10^3
P_5	7.4301×10^3	P_{10}	-3.6276×10^3

stress reduction in the bottom layer of the vulnerable component. It is evident that obtaining the minimum σ by theoretically calculating the optimal material parameters of the gasket is challenging. Therefore, we utilize the finite element calculation results from Section 3.1 to establish a fast prediction model for σ with different gasket materials through numerical fitting of surfaces. This provides a basis for the optimal selection of gasket materials.

Based on the previous analysis, the objective function for σ, which involves E and μ of the gasket, is set as Equation 8.

$$\sigma = \frac{p_1 + p_2 E + p_3 E^2 + p_4 E^3 + p_5 \mu}{1 + p_6 E + p_7 E^2 + p_8 E^3 + p_9 \mu + p_{10} \mu^2} \cdot \sigma_{eq} \qquad (8)$$

Among them, σ_{eq} represents the reference stress, which is the Von Mises stress of the bottom layer of the vulnerable component when $E = E_{eq}$. As discussed in Section 3.2.1, the stress at the bottom layer of the vulnerable component at this moment corresponds to an ideal cylindrical, resulting in $\sigma_{eq} = |\sigma_z|$. Hence, the theoretical Formula 3 is used to directly calculate σ_{eq}. P_1-P_{10} are all fitting coefficients. The Levenberg-Marquardt method is used for regression analysis fitting, and the parameter fitting results are shown in Table 7.

The fitted function surface is depicted in Figure 16. In the diagram, the x-axis and y-axis represent E and μ of the gasket, respectively, while the z-axis represents the corresponding Von Mises stress peak at the bottom layer of the vulnerable component. The blue sphere represents the simulation calculation result. The RMSE (root-mean-square error) of the final function is 0.0846, and the coefficient of determination R-Square is 0.99796, indicating that the fitting results is satisfactory.

To facilitate the optimal selection of gasket materials in practical design, we project Figure 16 upwards, obtaining a σ contour char, where E serves as the x-axis and μ as the y-axis, as shown in Figure 17. The area between the two yellow dashed lines in this figure represents the range of materials preferred in this article. Compared with the stress in the bottom layer of the vulnerable component under the traditional steel material bottom condition described in Section 2.4.2, the σ within the preferred interval are all below 3.24 MPa, showing a decrease of more than 34.4%. Reasonable selection of gasket materials effectively reduce the stress in the bottom layer of the vulnerable component, thus protecting the vulnerable component.

The black dots in Figure 17 are labeled with numbers, indicating the corresponding parameter positions for 20 traditional used materials or material types based on the statistical data presented

in this article (Sachin and Ravindra, 2012; Iqbal et al., 2013; Kou and Xu, 2015; Huang et al., 2010; Li et al., 2006; O'Daniel et al., 2002; Riedel et al., 2009; Urtiew et al., 2008; Hou and Werner, 1996; Mu et al., 2010; Kim et al., 2005; Ackland et al., 2013; Nsiampa et al., 2011; Fu et al., 2022). From the perspective of the capacity of various materials to reduce the peak Von Mises stress in the bottom layer of the vulnerable component, high polymer materials like engineering plastics generally outperform most metal and flexible materials. For example, Nylon 1,010, High-density polyethylene, Low-density polyethylene and Celluloid are all excellent materials in terms of performance. Of course, in the actual engineering design process, material researchers should comprehensively consider the vulnerable components to be protected and the actual load conditions.

4 Verification test

To validate whether gasket materials within the optimized parameter range specified in Section 3.3 effectively reduce the bottom layer stresses of the vulnerable energetic composites through homogenization-coordination mechanisms under low-amplitude, long-pulse dynamic loading conditions. This study uses simulated launch load testing methodology for validation experiments and selects process-stabilized Nylon 1010 as the preferred representative gasket material.

4.1 Test method selection

Common simulated launch load testing methods mainly include: closed bomb test, semi-closed bomb test, and large-scale drop hammer test (Liu et al., 2022; Xu and Qu, 2015; Zhou et al., 2023). However, it is difficult to measure stress distribution without interfering with stress concentration in vulnerable components in practical testing. Since one end of the specimen is typically constrained by ground anchorage, it remains challenging to determine through mechanical analysis whether the stress concentration point at the bottom of component initiates failure first. But for large-scale drop hammer tests, existing literature confirms that shear failure preferentially occurs at surface/interface stress concentration points in the selected vulnerable energetic composite components. The energetic properties induce localized friction and temperature rise during failure, collectively triggering decomposition behavior — a characteristic advantageous for observing and validating surface/interface protection (Baker et al., 1992; Wu and Huang, 2013; Duan et al., 2023). Accordingly, this study addresses the challenge of stress concentration testing at vulnerable material interfaces by utilizing the unique properties of the protected material, implementing comparative drop hammer testing to validate gasket performance.

4.2 Experimental setup

The large-scale drop hammer test employs a 400 kg drop hammer that free-falls along preset guide rails under gravitational force. The impact energy is transmitted via a piston to the head
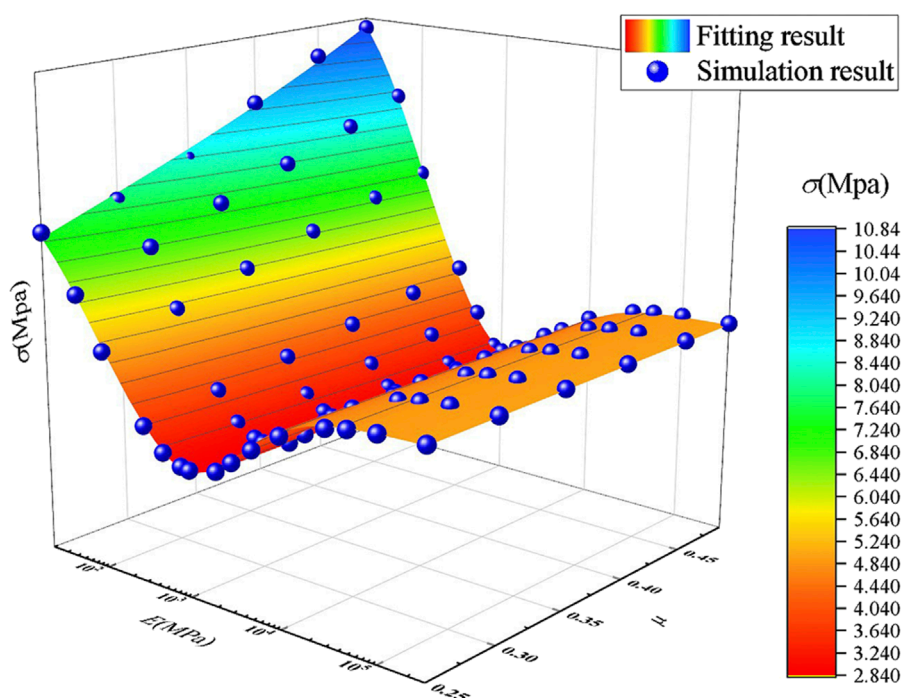
FIGURE 16
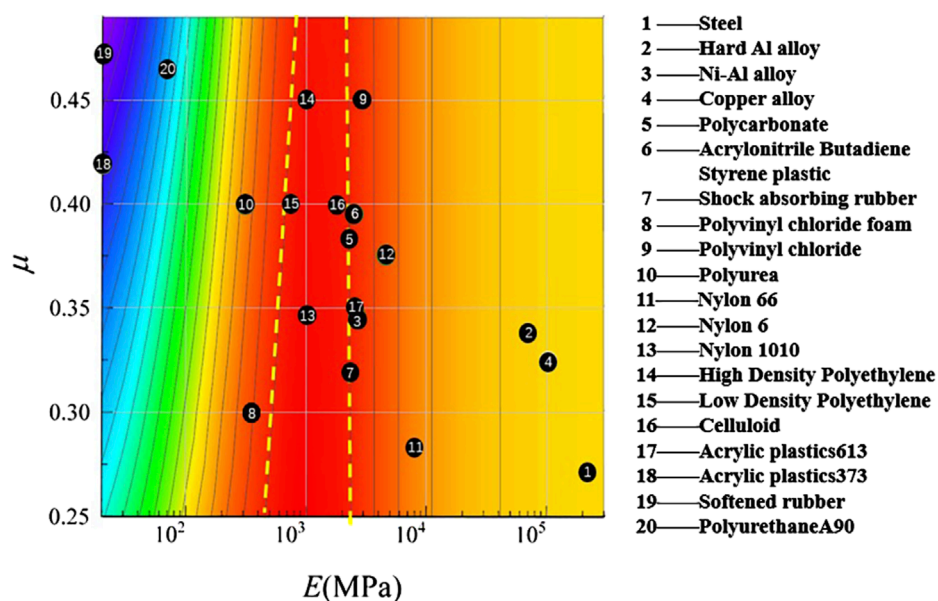Fitted function surface.

FIGURE 17
Performance projection chart of different gasket materials.

face of the sample within the shell, thereby simulating high-g axial loading conditions. Concurrently, the underlying data acquisition system measures and records the average axial stress at the sample's base surface with a sampling rate of 200,000 Hz. In large-scale drop hammer testing, the pulse duration of stress loading typically ranges from 3 to 10 ms, with peak stresses attainable up to 1,500 MPa and maximum stress rates achievable at 300 MPa·ms^{-1}. The impact loading characteristics closely approximate artillery bore pressure profiles, making this method extensively utilized in simulating launch load-induced shock. Figure 18 schematically depicts the large-scale drop hammer apparatus and its operational principle.

It is difficult to completely extract the test sample after testing without altering its original state. Therefore, the study first uses the 'up-and-down method' to determine the drop height corresponding to 100% decomposition reaction of the unprotected energetic composite material. Then, comparative tests are conducted on samples with gaskets under the same or even higher drop height conditions. Based on the preceding analysis of test methodologies, if the gasket-equipped samples can safely avoid decomposition reactions at higher drop heights, it can be concluded that the gasket protects typical vulnerable composite materials by reducing interface stress peaks.

The gasket material used in testing is Nylon 1010 (Wang and Huang, 1994; HG 2349, 1992; JB/ZQ 4196-2006, 2006), as shown in Figure 19a. As mentioned earlier, the thickness of the gasket in the preferred material range has almost no effect on the stress concentration and the peak Von Mises stress of the unprotected energetic composite material. Therefore, considering the convenience of experimental processing, the size of the gasket sample is taken as Φ 40 mm $\times$ 4 mm. During testing of samples containing gaskets, the gasket is bonded to the end face of the protected composite specimen using a two-component epoxy adhesive, as depicted in Figure 18. The protected vulnerable components are conventional energetic composites (Carlos and Scott, 2021), which are formed by compressing a mixture of

95.8 wt% 2,4,6,8,10,12-hexanitrohexaazaisowurtzitane (HNIW), 4.2 wt% plastic binder and graphite. Figure 19b shows 45 protected composite specimens manufactured under this formulation, designed as Φ40 mm $\times$ 40 mm cylinders with a density of 1.93 g·cm^{-3}. These specimens are labeled c-1, c-2, c-3, etc., for identification.

4.3 Experimental results and analysis

Based on gasket-free sample testing, impact-induced decomposition probabilities of vulnerable composite structures under 0.8–1.2 m drop height conditions were obtained as shown in Table 8. Through testing of samples with gaskets, decomposition probabilities of protected vulnerable structures under 1.0–1.2 m drop height conditions were acquired and presented in Table 9.

Test results demonstrate: Samples without gasket protection exhibited impact-induced decomposition reactions under all test conditions with drop heights $H \geq 1.1$ m. In contrast, samples equipped with Nylon 1,010 gaskets showed no impact-induced decomposition reactions at $H \leq 1.1$ m, with only small probabilities of reaction occurrence observed at $H = 1.2$ m.

For samples without gasket protection, measured stress-time profiles corresponding to critical drop height conditions in Table 8

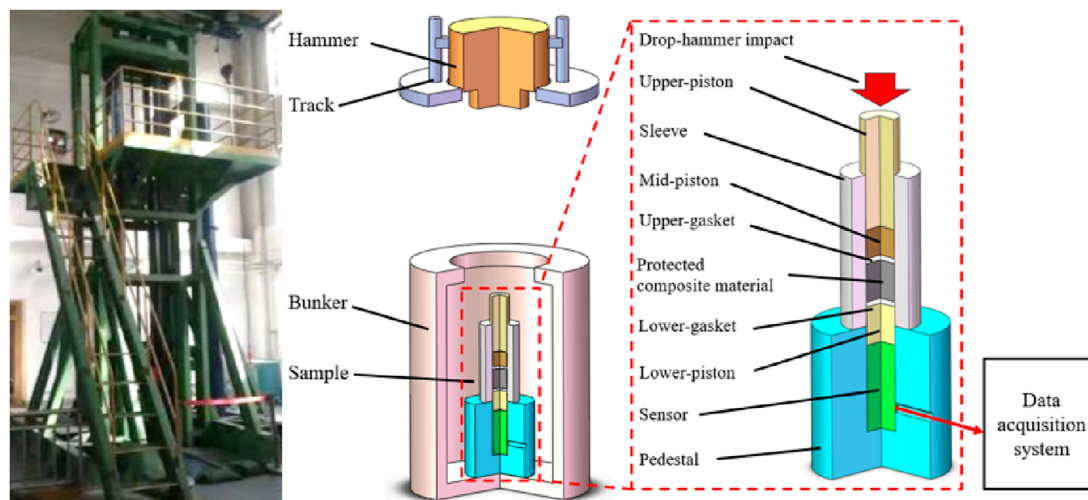FIGURE 18
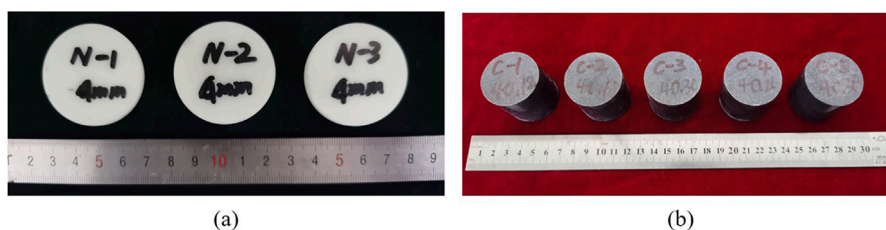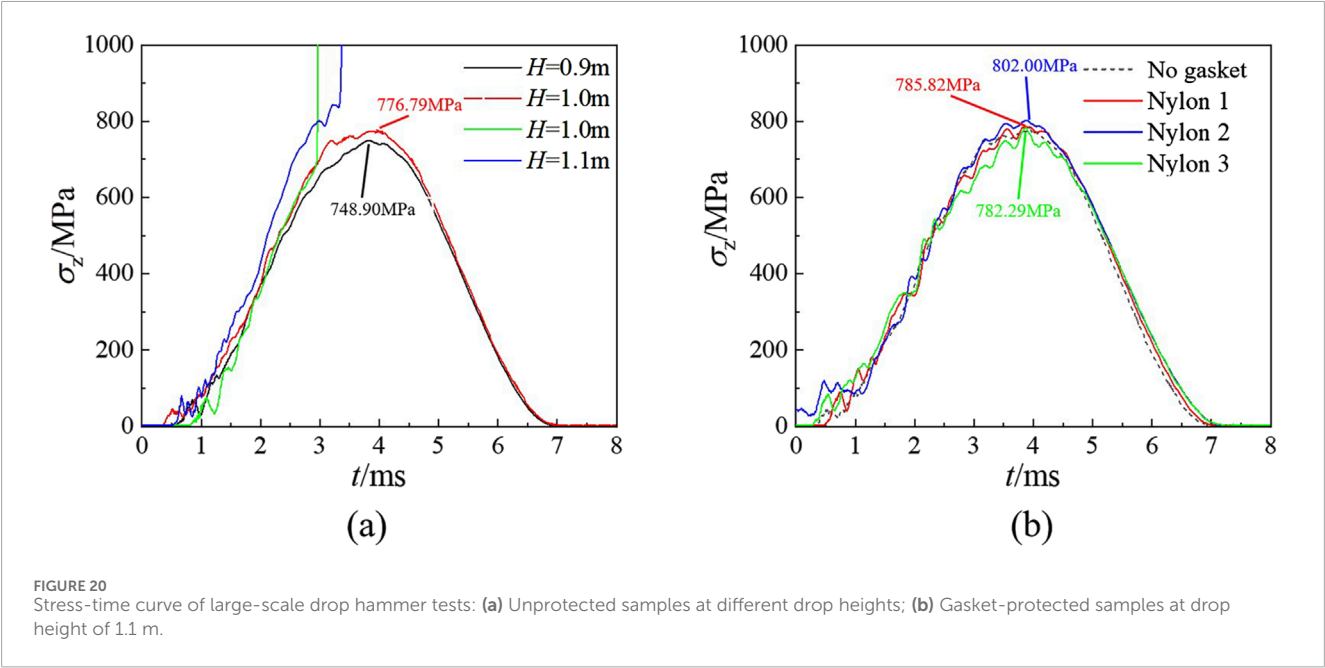Large-scale drop hammer device and its schematic diagram.

FIGURE 19
Photo of the test sample: **(a)** Nylon 1,010 gaskets; **(b)** Protected vulnerable components.

TABLE 8 Test results of drop hammer tests on gasket-free sample at different heights.

Drop height H (m)	Number of tests	Number of decompositions	Decomposition probability (%)
0.8	7	0	0
0.9	10	0	0
1.0	10	6	60
1.1	10	10	100
1.2	8	8	100

TABLE 9 Test results of drop hammer tests on samples with gaskets at different heights.

Drop height H (m)	Number of tests	Number of decompositions	Decomposition probability (%)
1.0	3	0	0
1.1	3	0	0
1.2	3	1	33.33

FIGURE 20
Stress-time curve of large-scale drop hammer tests: **(a)** Unprotected samples at different drop heights; **(b)** Gasket-protected samples at drop height of 1.1 m.

were extracted and plotted in Figure 20a. For drop heights of 0.9 m and 1 m, maximum stress curves under non-decomposition conditions were selected. Regarding gasket-equipped samples, three experimental stress-time profiles from the $H = 1.1$ m condition in Table 9 are shown in Figure 20b.

The experimental results demonstrate that the critical drop height threshold for impact-induced decomposition reactions in gasket-protected composite samples is significantly higher than in the unprotected state. Comparative analysis of the average axial stress on the surface of vulnerable samples in the drop hammer test reveals that the nylon gasket remains capable of effectively protecting the samples from impact-induced decomposition reactions, even under higher axial stress levels. Clearly, the gasket does not function by altering the actual magnitude of average stress, but rather improves the distribution of local stresses. This reduces the likelihood of localized thermal decomposition in energetic composites under simulated launch loads. The comparison of experimental results shows the optimally selected gasket material in this study indeed provides excellent protection for vulnerable composite materials subjected to low-amplitude, long-pulse dynamic loading conditions.

5 Conclusion

Through numerical simulation and large-scale drop hammer testing methodologies, this study investigates the impact protection mechanisms of engineering plastic gaskets on typical vulnerable components under low-amplitude, long-pulse dynamic loading conditions. The principal conclusions are as follows:

(1) The influence of gasket parameters on the stress in the bottom layer of the vulnerable component is presented. The σ_z and σ in the bottom layer of the vulnerable component with a gasket are influenced by the deformation capacity of the gasket material. Under the same μ, the σ value exhibits a trend of first decreasing and then increasing with the increase of the E of the gasket. This demonstrates a strong correlation with the magnitude of σ_z. For different μ, when the E of the gasket is small, its μ is positively correlated with σ and σ_z. When the E of the gasket is large, its μ has almost no effect on the σ and σ_z.

(2) The mechanism of the uniform and coordinated effect of the gasket on the bottom stress of the vulnerable component was revealed. The vulnerable component is protected under the launch load through deformation coordination by the gasket, which reduce the stress at the bottom of the vulnerable component, rather than through the mechanism of "buffering" or "attenuation". Under the given μ, the E of the gasket material has a deformation equilibrium point E_{eq}, which results in uniform distribution of σ and σ_z. At the same time, a stress coordination point E_{sc} is slightly greater than E_{eq}, which minimizes the peak σ under the combined action of various stresses. Both E_{eq} and E_{sc} have values that are negatively correlated with the μ of the gasket material. The disparity in deformation capabilities between the gasket and the vulnerable component leads to changes in the triaxial stress on the contact surface. This is the key to achieving stress homogenization and coordination in the bottom layer of the vulnerable component.

(3) The optimal parameter ranges for gasket materials are defined, and the protective effects of engineering plastic gaskets under low-amplitude, long-pulse dynamic loading conditions are experimentally validated. Materials within the recommended parameter ranges—such as Nylon 1,010, high-density polyethylene (HDPE), low-density polyethylene (LDPE), and celluloid — reduce the bottom layer von Mises stress in vulnerable components by over 34.4% compared with traditional metallic substrates. Additionally, using Nylon 1010 as the optimized material, comparative testing between unprotected and gasket-protected samples was conducted through large-scale drop hammer simulation experiments. This conclusively verifies the protective effectiveness of engineering plastic gaskets under low-amplitude, long-pulse dynamic loading conditions.

Conflict of interest

The authors declare that the research was conducted in the absence of any commercial or financial relationships that could be construed as a potential conflict of interest.

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References

Ackland, K., Anderson, C., and Ngo, T. D. (2013). Deformation of polyurea-coated steel plates under localised blast loading. *Int. J. impact Eng.* 51, 13–22. doi:10.1016/j.ijimpeng.2012.08.005

apps.dtic.mil (2025). Design accelerations for the army's excalibur projectile. Available online at: https://apps.dtic.mil/sti/tr/pdf/ADA435761.pdf/ (accessed on 15 April 2025).

Baker, P. J., Mellor, A. M., and Coffey, C. S. (1992). Critical impact initiation energies for three HTPB propellants. *J. Propuls. power* 8, 578–585. doi:10.2514/3.23517

Carlos, C., and Scott, I. J. (2021). Detonation performance of the CL-20-based explosive LX-19. *Proc. Combust. Inst.* 28, 3661–3669. doi:10.1016/j.proci.2020.07.089

Duan, H. Z., Wu, Y. Q., Yang, K., and Huang, F. L. (2023). Numerical simulation of drop weight impact sensitivity evaluation criteria for pressed PBXs. *Def. Technol.* 19, 76–90. doi:10.1016/j.dt.2022.01.013

Fu, J., Liu, G., Fan, C. H., Liu, Z. Y., and Luo, H. T. (2022). Design and experimental study on vibration reduction of an UAV lidar using rubber material. *Actuators* 11, 345. doi:10.3390/act11120345

HG 2349-1992. (1992). *Polyamide 1010 resin*. China Standards Press: Beijing, China.

Hou, X. F., and Werner, G. S. (1996). Projectile perforation of moving plates: experimental investigation. *Int. J. Impact Eng.* 18, 859–875. doi:10.1016/S0734-743X(96)00012-7

Huang, T. Y., Hong, Y., and Jin, S. L. (2010). Dynamic simulation of canister impact examination. *Mach. Des. Manuf. Eng.* 39, 74–76. doi:10.3969/j.issn.1672-1616.2010.05.019

Huang, Z. X. (2014). *Theory and practice of shaped charge*. 1st ed. Beijing: Beijing Institute of Technology Press, 216. Available online at: https://ss.zhizhen.com/detail_38502727e7500f2653114f3248dbb36576b35b9bb439a8a01921b0a3ea25510134114c96 9f2eae5cf859c3730be4d6288c64b13ae62da1f908e5d80a60b47bec3b971bc1e9c905f862 54471b25d14ada (accessed on April 15, 2025).

Iqbal, M. A., Khan, S. H., Ansari, R., and Gupta, N. K. (2013). Experimental and numerical studies of double-nosed projectile impact on aluminum plates. *Int. J. impact Eng.* 54, 232–245. doi:10.1016/j.ijimpeng.2012.11.007

JB/ZQ 4196-2006 (2006). *Nylon rods and tubes*. Beijing, China: China Standards Press.

Jia, S. X., Huang, M., Zhang, H., Hu, B. W., Yan, M., and Zhang, Y. D. (2023). Performance comparison of different materials grain cushion under high overload. *J. Proj. Rockets, Missiles Guid.* 43, 32–38. doi:10.15892/j.cnki.djzdxb.2023.04.005

Kim, H. S., Yeom, K. S., Kim, S. S., and Sotsky, L. (2005). "Numerical simulation for the front section effect of missile warhead on the target perforation," in *Proceedings of the 22nd international symposium on ballistics, Vancouver, Canada*. Available online at: https://www.researchgate.net/publication/313553392_Numerical_simulation_for_the_front_section_effect_of_missile_warhead_on_the_target_perforation (accessed on April 15, 2025).

Kou, J. F., and Xu, F. (2015). "The influence of bird's shape and attitude on bird-strike analysis of structure," in *Proceedings of the international conference on computer information systems and industrial applications (CISIA 2015), Bangkok, Thailand*. Available online at: https://libresource.bit.edu.cn/https/443/cn/clarivate/webofscience/yitlink/wos/alldb/full-record/WOS:00035986 6200267 (accessed on April 15, 2025).

Li, D. W., Tan, Z. J., Hao, Y., Zeng, L. Q., and Wang, H. (2021). Effect of cushion materials and structure on protection effect of warhead charge. *J. Ordnance Equip. Eng.* 42, 36–41. doi:10.11809/bqzbgcxb2021.10.006

Li, Y. Y., Jia, X. Z., Gao, L. L., and Miao, X. (2017). Numerical simulation on cushion property of projectile inner cushion material during penetration. *New Chem. Mater.* 45, 128–130. Available online at: https://kns.cnki.net/nzkhtml/xmlRead/trialRead.html?dbCode=CJFD&tableName=CJFDTOTAL&fileName=HGXC201703044&fileSourceType=1&invoice=rZwY1U3N pNU5SuNISSVkHjfNH9z2enGpDpCu1Ml6poKpTCSOxgZpGQXUgvnf0%2buITb5B mVf5WU8dZIE0K9h%2fv6c5vg2YGiwMK8f%2f2slDGuUsI3BZmkEpyvToNEXQGsg DqIpLozFWfYrGzSFXcMQC2BE3b92OdjHJpi40Ez7FnF8%3d&appId=KNS_BASIC_PSMC (accessed on April 15, 2025).

Li, Z., You, M., and Kong, F. R. (2006). Numerical simulation for cleavage joints of two kinds of adhesives based on ANSYS. *Chem. Adhesion* 5, 299–301. doi:10.3969/j.issn.1001-0017.2006.05.004

Liang, F. D., Liang, Z. Y., and Deng, D. Z. (2020). Study on the composite structure of aluminum foam-filled thin-walled metal tube to reduce the charge overload

inside the projectile during the penetration process. *Shock and Vib.* 2020, 1–11. doi:10.1155/2020/8887893

Liu, W., Wang, G. P., Rui, X. T., Gu, J., and Zhao, X. (2022). A hotspot model for PBX explosive charge ignition in a launch environment. *Combust. Sci. Technol.* 194, 1954–1972. doi:10.1080/00102202.2020.1849166

Liu, W., Wang, G. P., Rui, X. T., Li, C., and Wang, Y. (2024b). A test method for launch safety of explosive charge accurately simulating launch overload. *J. Energetic Mater.* 42, 479–499. doi:10.1080/07370652.2022.2108165

Liu, Y. R., Liu, Z. P., Ji, C. C., Yuan, D. X., and Jiang, L. (2024a). Experimental and numerical study on cushioning and energy absorption properties of artificial cartilage biomimetic material. *Aerosp. Mater. and Technol.* 54, 24–30. doi:10.12044/j.issn.1007-2330.2024.06.003

Mohammad, N. D., and Mohammadzadeh Gonabadi, A. (2019). Numerical study of energy absorption in aluminum foam sandwich panel structures using drop hammer test. *J. Sandw. Struct. and Mater.* 21, 3–18. doi:10.1177/1099636216685315

Mu, Y., Zhao, G. Q., and Zhang, C. R. (2010). Numerical investigation of die geometry effect on LDPE annular extrudate swell. *J. Appl. Polym. Sci.* 117, 91–109. doi:10.1002/app.31490

Nsiampa, N. N., Robbe, I. C., Papy, I. A., and Dr, I. (2011). "Development of a thorax finite element model for thoracic injury assessment," in *Proceedings of the 8th European LS-DYNA conference, strasbourg, France*. Available online at: https://www.dynamore.de/de/download/papers/konferenz11/papers/session7-paper4.pdf (accessed on April 15, 2025).

O'Daniel, J. L., Krauthammer, T., Koudela, K. L., and Strait, L. H. (2002). An UNDEX response validation methodology. *Int. J. Impact Eng.* 27, 919–937. doi:10.1016/S0734-743X(02)00014-3

Riedel, W., Kawai, N., and Kondo, K. (2009). Numerical assessment for impact strength measurements in concrete materials. *Int. J. Impact Eng.* 36, 283–293. doi:10.1016/j.ijimpeng.2007.12.012

Sachin, S. G., and Ravindra, K. S. (2012). A numerical study on effect of strain rate and temperature in the taylor rod impact problem. *Int. J. Struct. Changes Solids* 4, 1–11. Available online at: https://www.researchgate.net/publication/259860307_A_numerical_study_on_effect_of_strain_rate_and_temperature_in_the_Taylor_rod_impact_problem (accessed on April 15, 2025).

Sun, J., Zhao, C. X., Li, J., Mei, H., Liu, X., and Yan, S. L. (2025). The fracture modes of biomimetic borosilicate glass protective composite. *Materials* 18, 739. doi:10.3390/ma18040739

Urtiew, P. A., Vandersall, K. S., Tarver, C. M., Garcia, F., and Forbes, J. W. (2008). Shock initiation of composition B and C-4 explosives: experiments and modeling. *Russ. J. Phys. Chem. B Focus Phys.* 2, 162–171. doi:10.1134/S1990793108020036

Wang, M. S., and Huang, J. C. (1994). Nylon 1010 properties and applications. *J. Polym. Eng.* 13, 155–174. doi:10.1515/POLYENG.1994.13.2.155

Wu, Y. Q., and Huang, F. L. (2013). Experimental investigations on a layer of HMX explosive crystals in response to drop-weight impact. *Combust. Sci. and Technol.* 185, 269–292. doi:10.1080/00102202.2012.716108

Xu, Z. F., and Qu, K. P. (2015). Study on experimental method of simulation loading for launch safety of charge. *Initiators and Pyrotech.* 6, 51–53. doi:10.3969/j.issn.1003-1480.2015.06.014

Zhang, Y. G., Lou, J. F., Zhou, T. T., Hong, T., and Zhang, S. D. (2016). Initial study on constitutive model of PBXs *via* viscoelastic statistical crack mechanics including anisotropic damage. *Chin. J. High Press. Phys.* 30, 301–310. doi:10.11858/gywlxb.2016.04.006

Zhou, L., Ni, L., Li, D. W., Zhang, X. R., Liu, H. Q., Jiang, T., et al. (2023). Test method for anti-overload performance of explosives. *Acta Armamentarii* 44, 1722–1732. doi:10.12382/bgxb.2022.0074